# A Significant Update to the Hyper-Modular Approach to Space Solar Power SPS-ALPHA Mark-IV

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# Space Solar Power : How IT WOULD WORK



#### THE SUN

- Can power 2,880 trillion light bulbs
- 1.4 million kilometer diameter
- The Sun has enough hydrogen fuel for billions of years



#### SPS-ALPHA SPACE-BASED HARVESTING

- ~6 km reflector array
- ~1.8 km solar PV panels + wireless power transmitter array
- ~7 km backbone structure
- Modular, robotic construction
- Cheap to launch; less than \$1,000/kg
- 99.95% Available Power

#### MICROWAVE ENERGY TRANSFER

- Precisely controlled transmission of energy
- Less than 20% of summer sunlight
- Can be "shared" across receivers and coordinated with ground-based solar



#### **GROUND STATION**

- ~6km diameter (elevated 5-10 m)
- Outside metro areas
- Mesh RF 'Rectifying Antenna' system
- Uses batteries to modulate supply to the existing electricity grid



#### EXISTING INFRASTRUCTURE

- DC or AC fed into the local grid
- Resembles
  Hydroelectric
  Power but...
- "Always" available
- "Shareable" across markets



#### HOMES AND BUSINESSES

- Base Load low cost
  electricity
- No carbon emissions
- Supports use at all hours of the day

#### 2024 SSP Modeling Generic Functional Work Breakdown Structure (WBS) for SSP (Including the Space Segment)



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# Major Options for SPS (Design, Technology, CONOPs...)

- Critical Characteristics for Solar Power Satellite Platforms
  - Frequency / type of power delivery from space to ground : WPT or ?
  - Voltage/Scale of the power management and distribution (PMAD) system
  - o Use of rotary systems: with / without PMAD? At What Scale?
  - Active thermal or not?
  - Type of structural system: 'stick built'? large modular? thin-film?
  - Robotic assembly or kinematically deployed structural systems?
  - Type of solar power generation (SPG): PV, dynamic, solar-pumped, mirrors?
  - SPG input: solar redirection using Reflectors or not? Large single mirror or smaller heliostats?
  - One platform or more? Physically connected or not?

Evaluation of Concepts & "Programmatics" Key Issues for Scaling from Experiments to Demonstrations to Systems

### **KEY SPS DESIGN ELEMENTS**

- Optical Systems
- Power Generation Systems
- Power Management & Wiring
- Thermal Management
- Structures & Mechanisms
- Wireless Power Transmission Components
- Robotics
- Intelligent Systems & Security (S/W & H/W)
- Attitude Control & Propulsion
- ETO and In-Space Transport

# SCALABILITY FACTORS

- Modularity
  - Size of "Modules"
  - Choices between "Bigger Modules" vs "More Modules"
  - "Connecting Tissue Issues"
- Manufacturability of Modules
- Control & Communications
- Construction, Evolution, Adaptability & Reconfigurability

### METRICS

- Cost (R&D, FOAK, NOAK)
- TRRA Assessment (TRL, TNV, R&D<sup>3</sup>)
- MRL Assessment
- SW-TRL Assessment

How Might Space Solar Power be Accomplished? The Hyper-Modular Architectural Approach

# Complex, "hyper-modular" architectures found in Nature...

Single genetic "individuals" comprising thousands to tens of thousands of "modules"

- Example: Ants capable of forming structural systems from themselves
- Example: Bees capable of navigation, cooperation and construction

Diverse genetic "individuals" in a single community comprising 100s of species and many 1000's of individuals

#### Introduced as a Space Solar Power Concept in 2012 during the NIAC SPS-ALPHA Study, Hyper-modular SPS with Robotic Assembly emerged as <u>the</u> most promising approach to SSP...





# Evolution of SPS-ALPHA

SPS-ALPHA Mk-0 (2011)

SPS-ALPHA Mk-I (2012)





SPS-ALPHA Mk-III (2020)



### SPS-ALPHA Mk-IV (Solar Power Satellite VIA Arbitrarily Large Phased Array)



## Comparison of SPS-ALPHA Mark-III and Mark-IV

# SPS-ALPHA Mark-III

Operational Orbit : GEO WPT Frequency : ~2.45 GHz Power Delivered : ~2 GW Mass : ~7,600 MT # Modules : ~1.9M Installed Cost : ~\$9.8 M ETO Cost : ~ \$200 / kg Rcvr Diameter : 6.2 km



Mass : ~3,200 MT # Modules : ~800K Installed Cost : ~ \$4.9 B ETO Cost : ~ \$200 / kg

Rcvr Diameter : 3.6 km

# A Roadmap to Operational / Cost-Effective SPS-ALPHA Mk-IV within a Decade









Phase 1

#### System-Level Lab Prototype @ ~50kW ~\$40M 12-18 months

Phase 2

System Demonstration in LEO @ ~300kW ~\$250M +18-24 months Phase 3

MEO Pilot Plant @ 10-100MW ~\$1B +24-36 months (to be updated) Phase 4 + Operational SPS in GEO @ 1-2GW +\$4B-\$6B +36-60 months,

4/17/24

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